

# Fracture Toughness Testing

# Organization for Expts. 5 and 7 (weeks of 4/2 and 4/16)

## Experiment 5

- Lab in 127 Norris
- Can visit lab in advance Monday 4/2 and Monday 4/16 from 9am-12pm
- Submit logbook preparation to TA in advance as normal
- Two teams working together to test three samples. Lab is over when samples tested and logbooks complete (may not take full lab period).
- Your regular TA will be there.
- Each team submits their completed logbook to their TA before leaving lab

Don't forget laptops

## Experiment 7

- Lab in Stability Tunnel
- Can visit in advance Friday 3/30 and Friday 4/13 from 8am to 12pm and 1pm to 5pm.
- Submit logbook preparation to TA in advance as normal
- First team\* arrives at start of lab period. Has 75 mins to complete test. Second team arrives 90 minutes after start of lab, has 75 mins for test.
- Regular TA not there. Submit logbook to wind tunnel operator before leaving lab

\* First team is the team with the lowest number on the experiment schedule

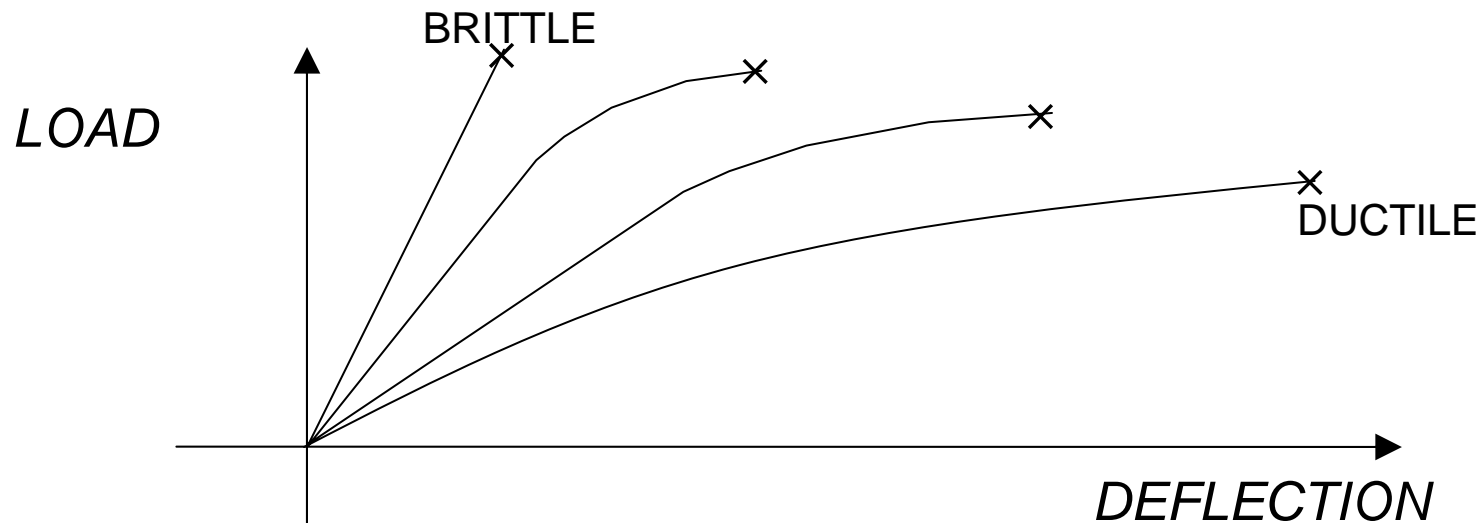
# Organization for 5<sup>th</sup> Instrumentation Lab Period (week of 4/9)

- Lab in Randolph 25 (next to the open-jet tunnel)
- Apply digital data acquisition and processing, including your own LabView programs, to the dynamic beam structure of Experiment 6
- Works just like a regular experiment
  - Read manual (chapter 5 of instrumentation lab)
  - Meet with your team in advance
  - Visit the lab
  - Do a logbook preparation
  - Logbook submitted and end of lab is graded
- Note your logbook preparation should be submitted to Dustin Grissom (dgrissom@vt.edu) not your regular TA.

Don't forget laptops

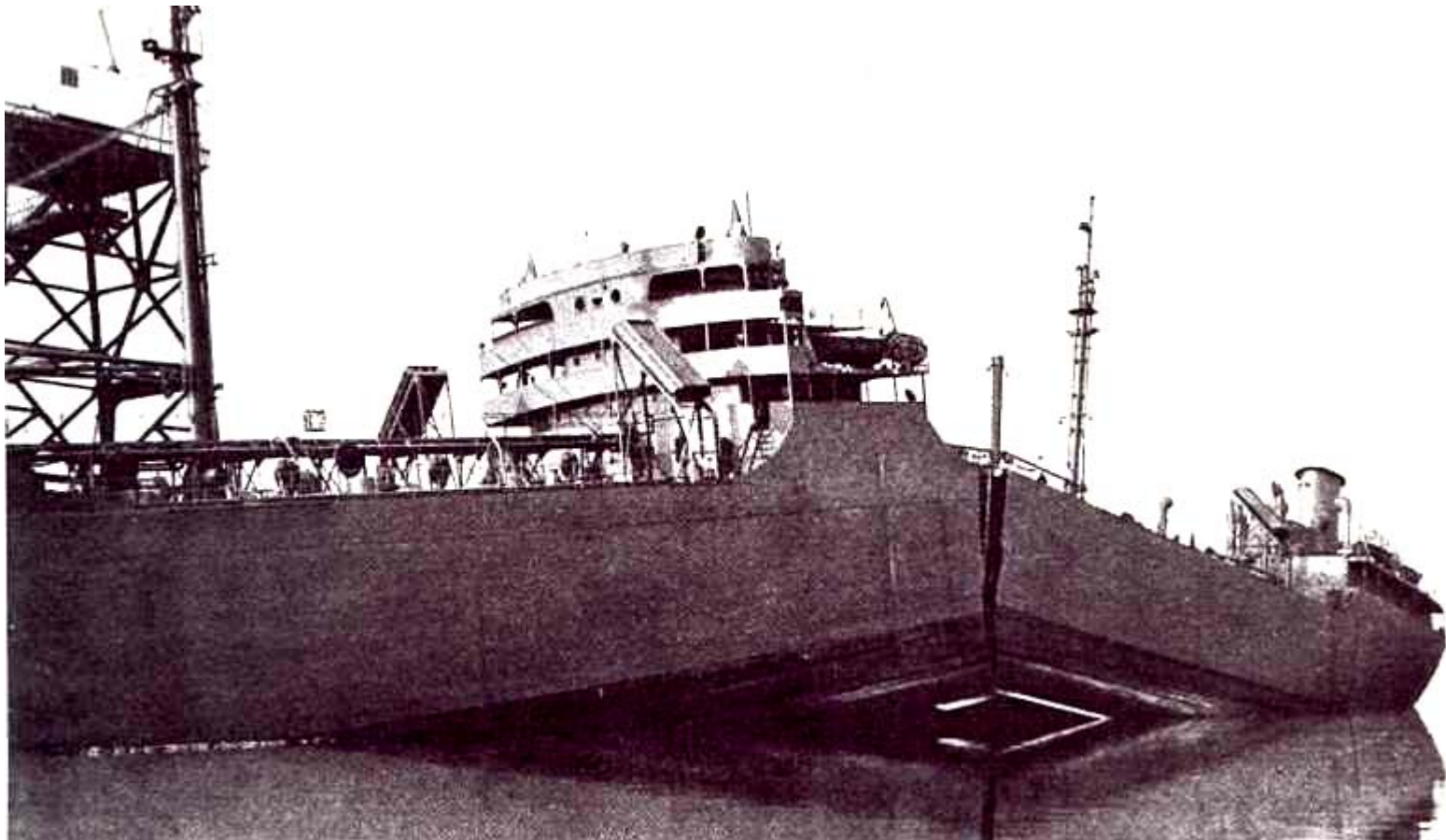
# Fracture

- Breaking of structural components into two or more parts
- Brittle fracture - low energy absorption
- Ductile fracture - large energy absorption

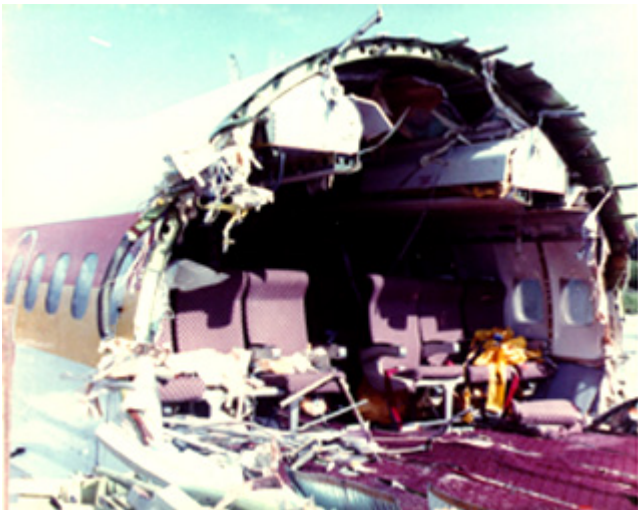


# Tanker SS Schenectady

(24 hours after launch)



# Aloha Airlines



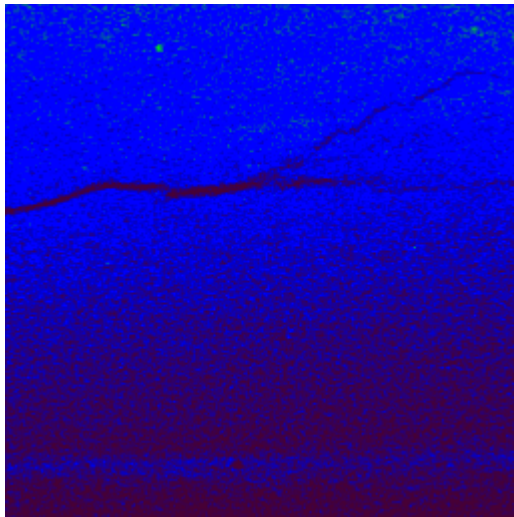
# Aloha Airlines



# Why does fracture occur?

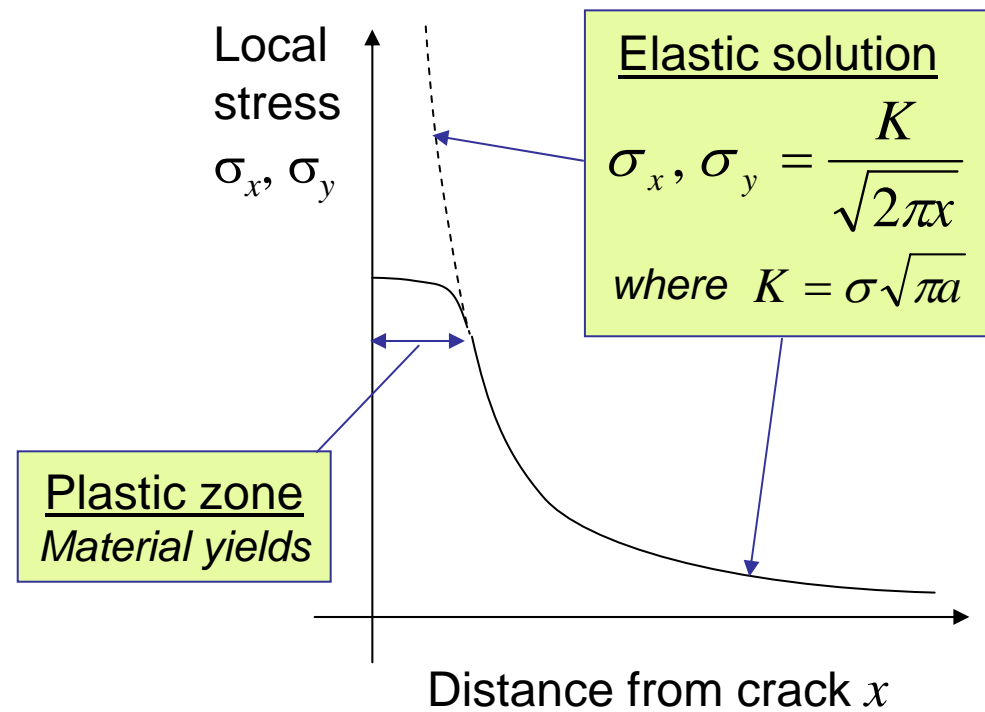
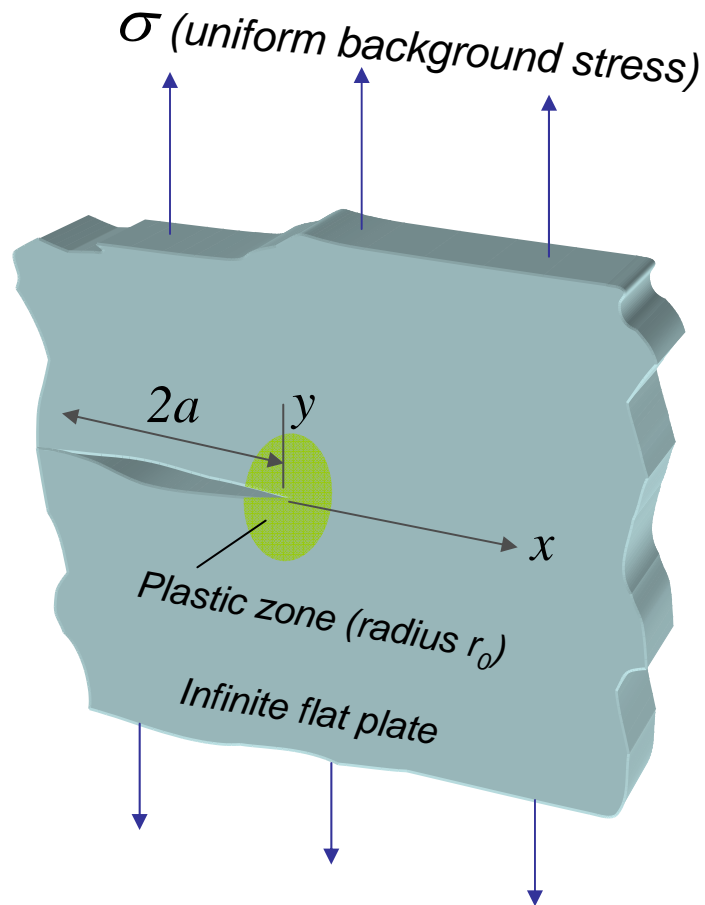
- Load increases to a point where cracks grow catastrophically.
- There are always cracks
  - They form as part of the manufacturing
  - They develop over time as a result of fatigue
- The strength of materials in the presence of cracks is therefore critical in defining when they will fail.

# F-111 Crack



# The Ideal Crack

*The Linear Elastic Fracture Mechanics (LEFM) Approach*

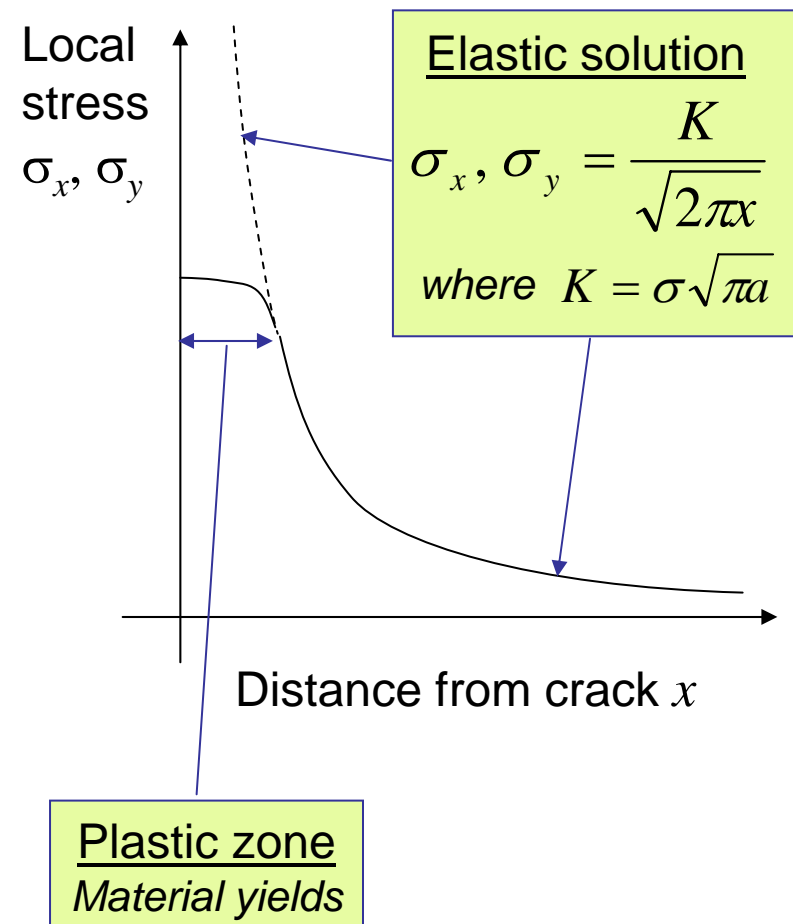


Even though the crack is embedded in a plastic zone, it is the elastic solution, in particular  $K$  that determines the stress field through which the crack would advance

# Stress Intensity Factor $K$

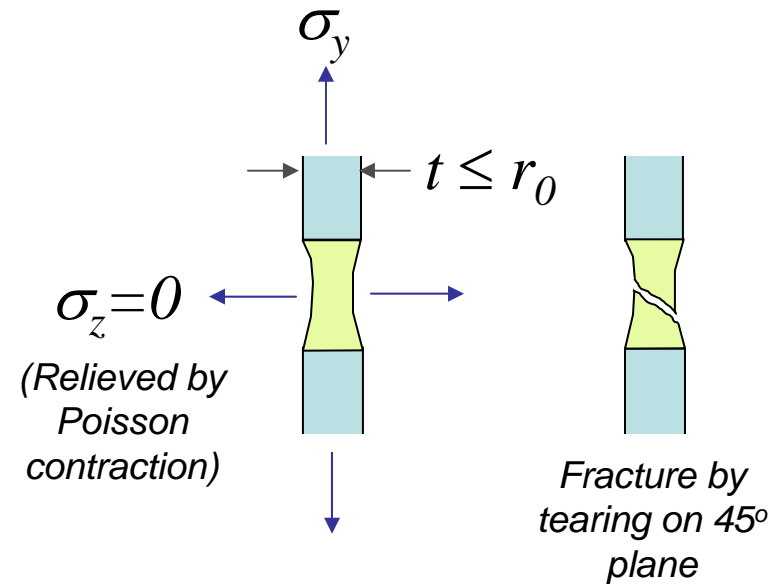
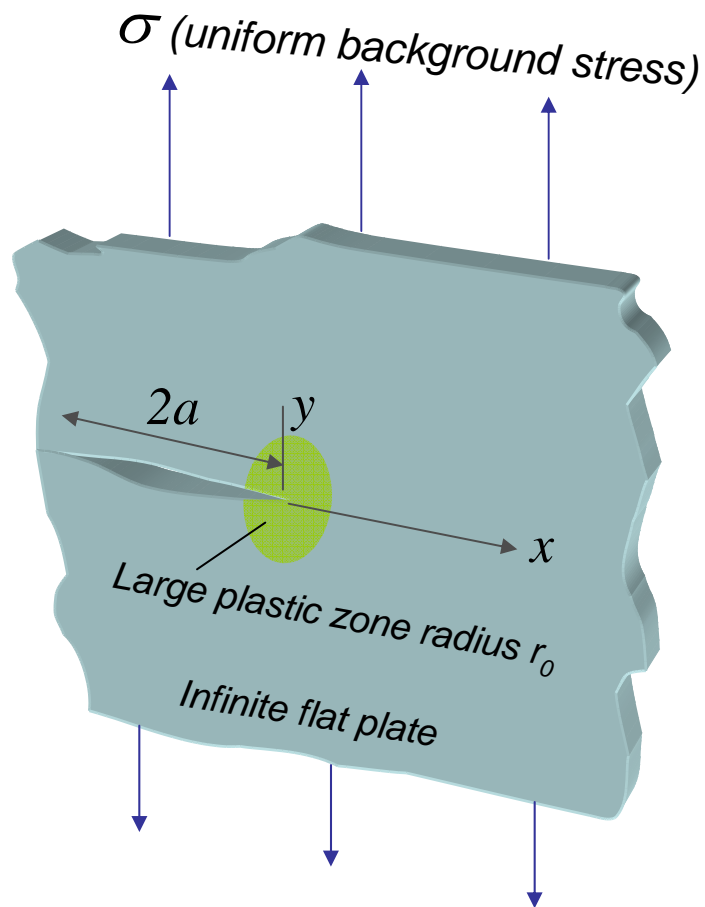
*The Linear Elastic Fracture Mechanics (LEFM) Approach*

- Scale of the elastic stress field generated by the crack
- Units of
  - Pa√m (usually MPa√m)
  - p.s.i.√in (usually k.s.i.√in)
- The stress intensity when the crack advances catastrophically is a measure of the strength of the material in the presence of a crack.
- This is called the Fracture Toughness  $K_c$



# Effect of Thickness

## Thin Plate

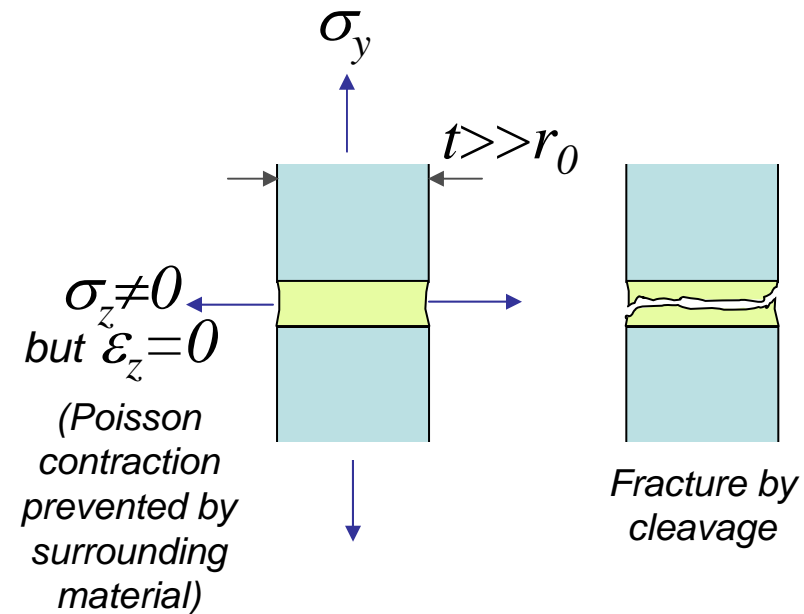
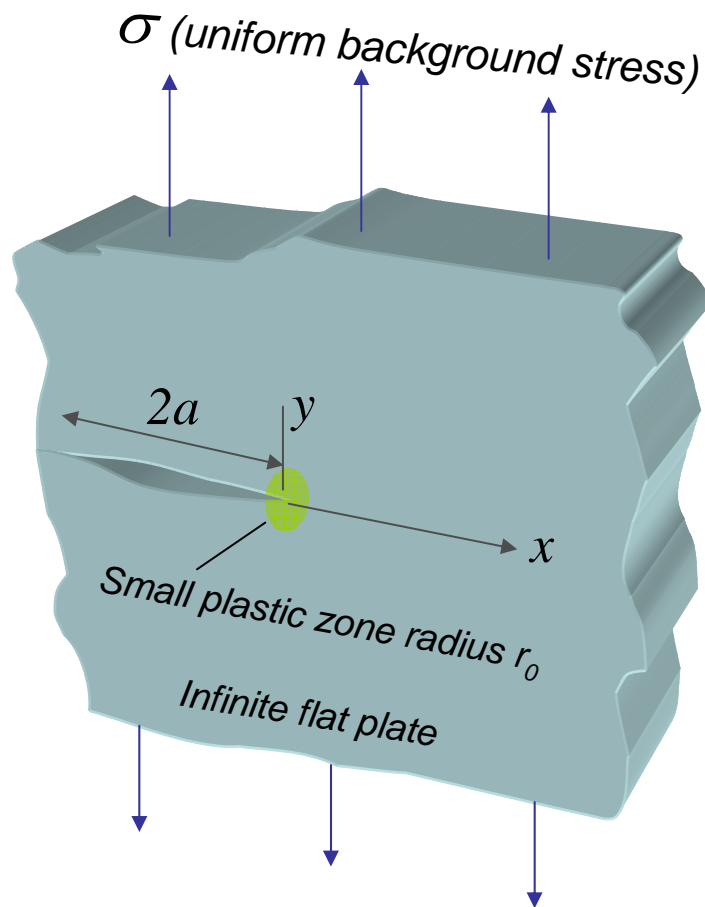


Deformation in z results in:

- stresses in only x and y - **Plane Stress**
- relief of some of the stress around crack
- a large fracture toughness, that depends on thickness
- ductile fracture takes place by tearing on  $45^\circ$  plane

# Effect of Thickness

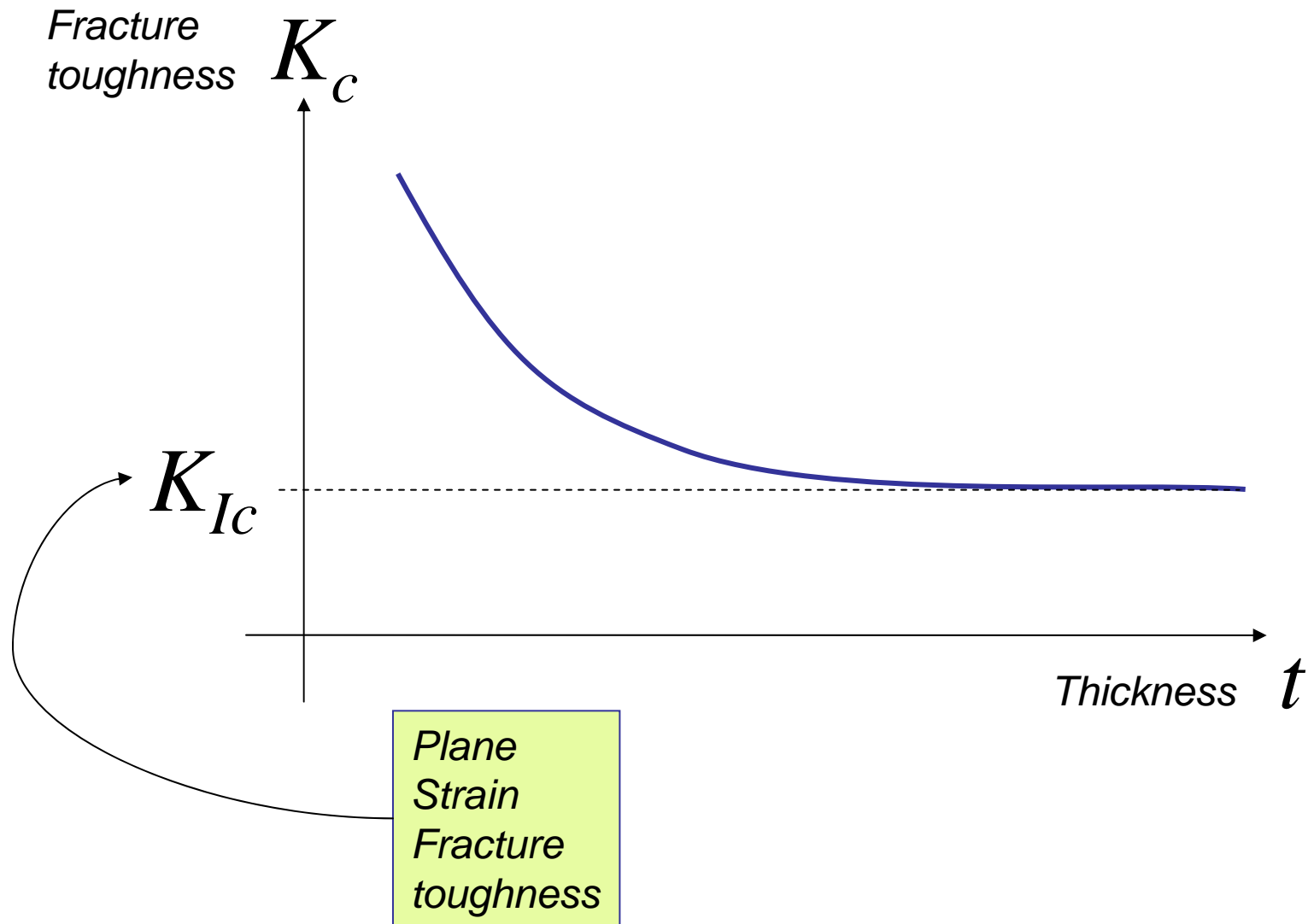
## Thick Plate



Lack of deformation in  $z$  results in:

- strain in only  $x$  and  $y$  - **Plane Strain**
- no relief of stress around crack
- a lower fracture toughness, that is independent of thickness
- brittle fracture on horizontal plane

# Effect of Thickness



# $K_{Ic}$ for different materials



material		yield	toughness
		MPa	MPa√m
Aluminium	2024-T851	455	26
	7075-T651	495	24
Titanium	Ti-6Al-4V	910	115
	*Ti-6Al-4V	1035	55
Steel	4340	860	99
	*4340	1515	60
	52100	2070	14

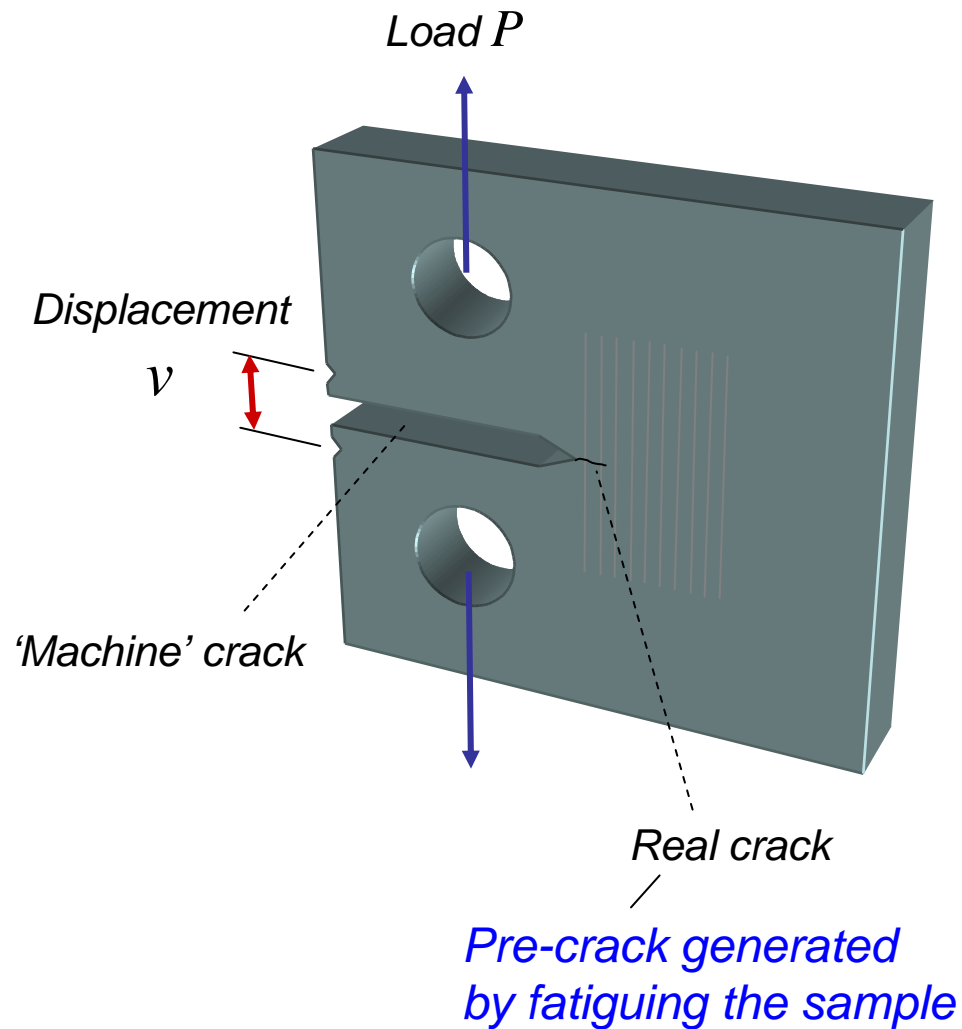
\* heat treated for higher strength



# Experiment 5

- Objective
  - To measure this material property, the plain strain fracture toughness, for Aluminum
- Approach
  - Break samples of different thickness using fracture toughness testing procedures specified in ASTM Standard No. E399
- Organization
  - Two teams of students have 3 samples between them
- Location
  - ESM Materials Lab, Norris 127

# Samples



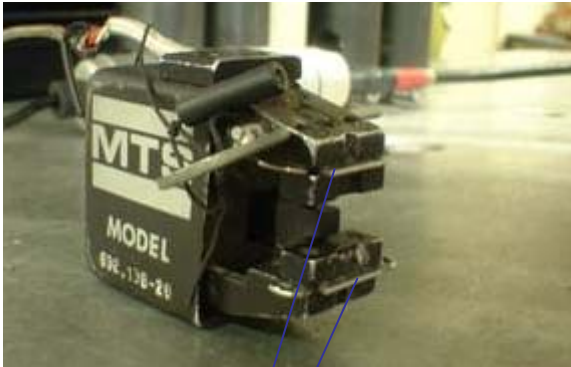
- Measure displacement vs. load until sample breaks.
- Identify load at fracture  $P_Q$
- Use this to determine fracture toughness



# Load and Displacement

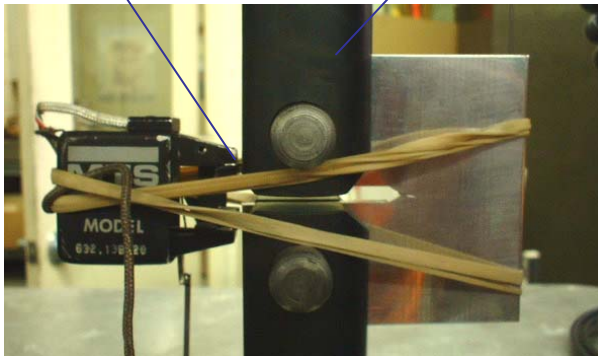
## Extensometer

(senses crack opening displacement)



*Jaws fit in notches in side of sample*

*Clevis grips hold sample in machine*



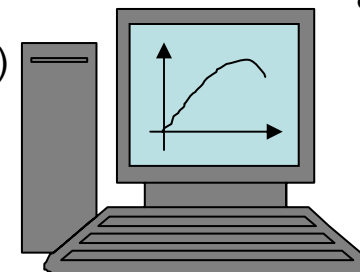
## Testing machine

(applies displacement, senses force)



## Computer with LabView

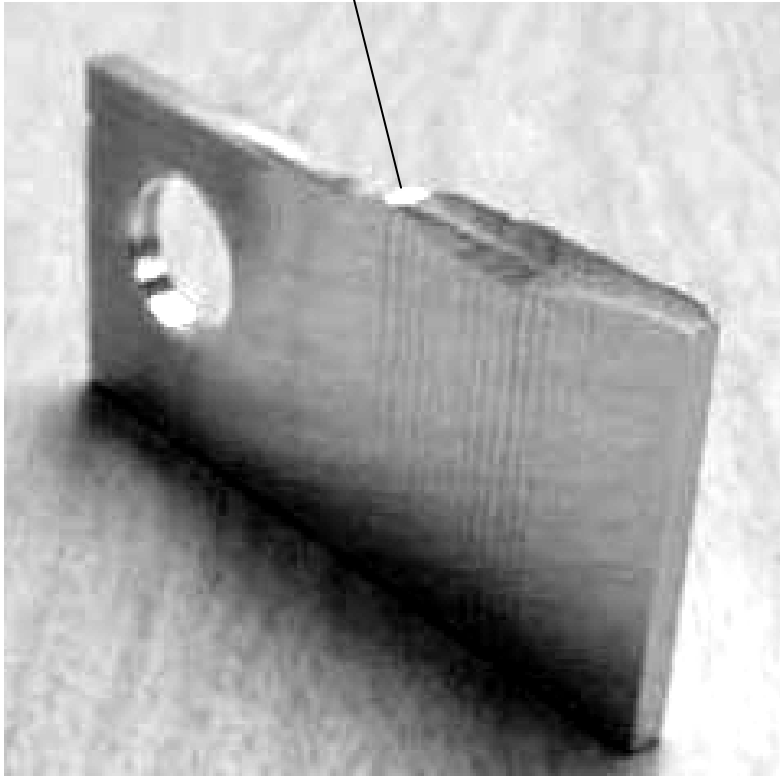
(reads and records load and displacement)



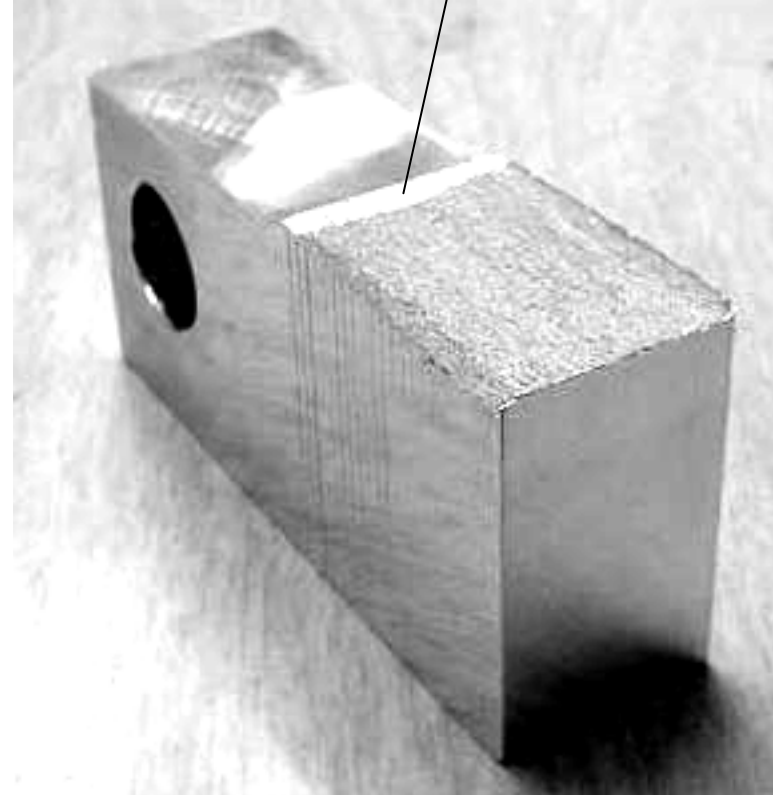
Controller and A/D

# Broken Samples

*Pre-crack*



*Pre-crack*

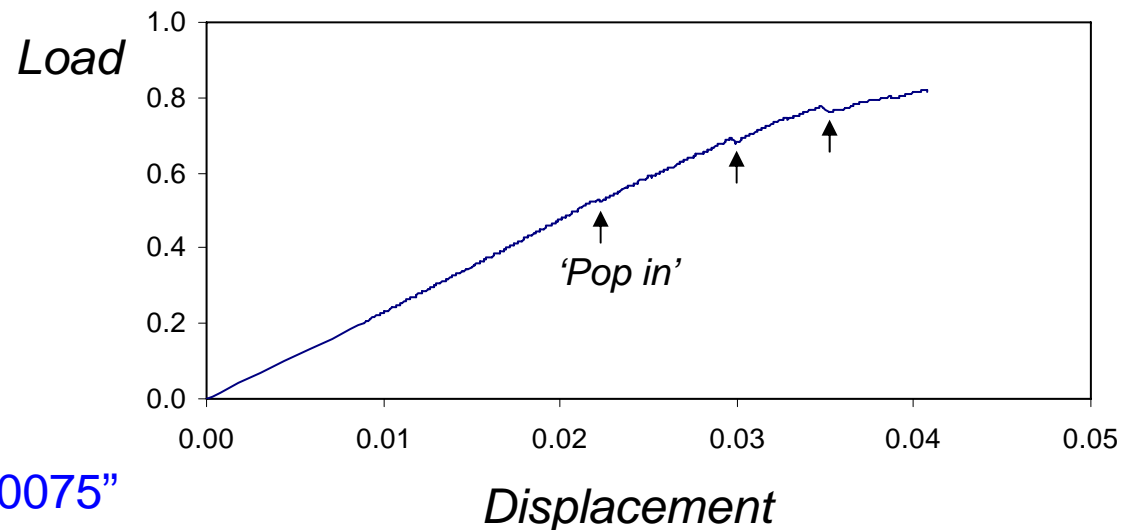
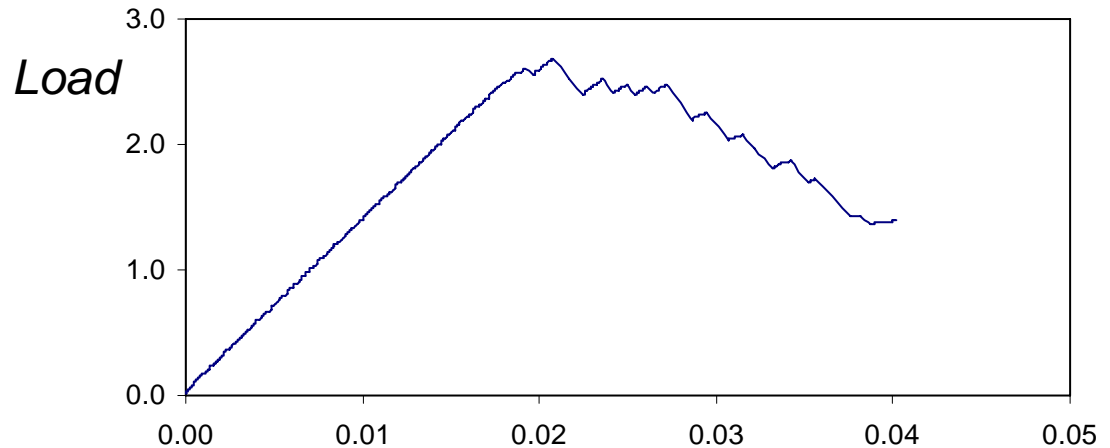


# Load at Fracture?

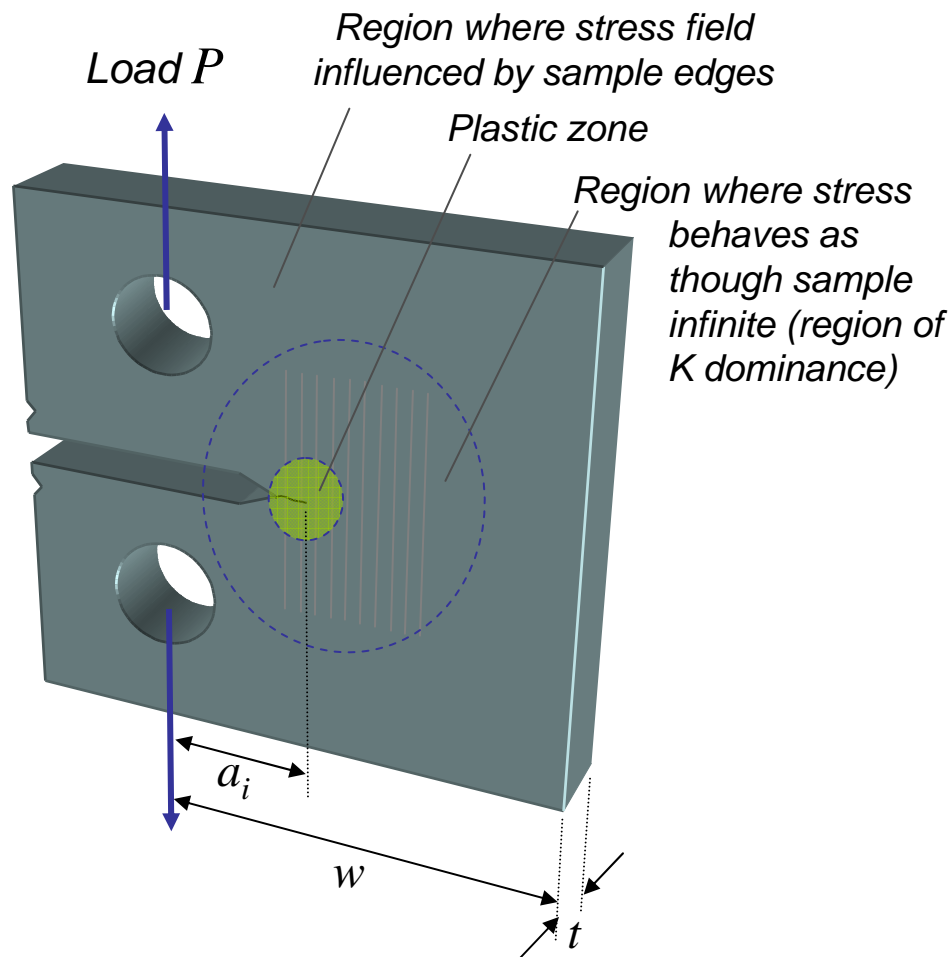
- Ductile or brittle?
- Elastic or plastic?
- Note that deciding when fracture occurs involves some choices.

Loadcell range 10 kips  
Extensometer range 0.075"

Manufacturer gives:  
Loadcell accuracy 10lbs  
Extensometer accuracy 0.000075"



# Fracture Toughness?



- Real sample is finite and applied stress is not uniform
- However, if plastic zone is small enough then a region exists where stress field behaves as though sample is infinite.
- The stress intensity in this region is a function of the applied load and the sample geometry. This function is known in the form of a curve fit, derived from finite element analysis.

$$K = \frac{P}{t\sqrt{w}} f\left(\frac{a_i}{w}\right)$$

$$\text{In } \infty \text{ case } K = \sigma\sqrt{\pi a}$$

# Ex. 5 Summary

- Pre-defined goal and procedures
- Two teams collaborate to break 3 samples
- Measure/photo samples before and after fracture
- Analyze the load vs displacement curves and the sample dimensions are analyzed to yield
  - The fracture toughness
  - The plastic zone size
  - Whether the sample is in plane stress or plane strain
- When you have completed all analysis, and have a plot of fracture toughness vs. sample thickness you are done.