Aircraft Noise

by

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NOMENCLATURE

C	Tone Correction Factor
d	Time interval during which $PNLT(k)$ is greater than $PNLTM - h$
D	Duration Correction Factor
<i>EPNL</i>	Effective Perceived Noise Level
F	Difference between Original and final Background Sound Pressure Level
h	The level to be subtracted from <i>PNLTM</i> that defines the duration of noise.
i	Frequency band indicator
k	Time indicator
n	Annoyance value
N	Combined Annoyance value over Frequency
P	Sound Pressure
P_{ref}	Reference Sound Pressure
PNL	Perceived Noise Level
PNLT	Tone Corrected Perceived Noise Level
PNLTM	Maximum Tone Corrected Perceived Noise Level
s and s'	Sound Pressure Level differences between frequency bands
\overline{s}	Average differences in Sound Pressure Levels
SPL	Sound Pressure Level

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SPL' and SPL"	Average Sound Pressure Levels
t	Time
dt	Length of time increments
T	Normalizing time constant

ACRONYMS AND ABBREVIATIONS

ANOPP Aircraft Noise Prediction Program

ACSYNT Aircraft Synthesis Program

DOT Department of Transportation

FAA Federal Aviation Agency

FAR Federal Aviation Regulations

FLOPS Flight Optimization System

FOOTPR Aircraft noise prediction program by Bruce Clark

HSCT High Speed Civil Transport

ICAO International Commercial Aviation Organization

NASA National Aeronautics and Space Administration

1. INTRODUCTION

The rapid spread of civil aviation, from the early elite of the 1960s to the tourists of today has exposed communities to a noise intrusion with unsociable levels and frequencies. With the introduction of the bypass and turbofan engines in the 1960s and 1970s and the subsequent evolution of these two engine technologies, aircraft have become progressively quieter, although the reduction in noise levels has not sufficiently offset the increase in operations or their psychological impact.

It is the purpose of this paper to derive and explain the procedure of obtaining a single indicator of this psychological effect, however, with the realization that not all factors can be included. Each person reacts differently to noise exposure depending on his or her tolerance level. This is furthermore only for a single noise event, studies have shown that the number of events can have even greater impact than the loudness of the noise itself.

Allowable levels of noise exposure and certification schemes are then outlined as dictated by the Federal Aviation Agency.

This is a partial fulfillment of a research effort on the community noise problem for HSCT aircraft.

2. NOISE CALCULATIONS

The certification criteria in FAR part 36 consists of meeting specified noise levels at certain reference points. The noise evaluation is designated *EPNL* (Effective Perceived Noise Level) and is measured in units of EPNLdB. This is a single number indicator of the subjective effects of airplane noise on human beings.

2.1 Basic Properties

Noise, or undesirable sound, is a physical phenomenon detected by humans as sound waves, which are only minute fluctuations in atmospheric pressure. This pressure is usually reported as *SPL* or sound pressure level, and since human hearing has adopted to million fold differences in sound pressure it is measured in units of decibels, which is a logarithmic scale. If the sound pressure is given in other units, it can be transformed to decibels using the following equation:

$$SPL = 20 \log_{10} \left(\frac{P}{P_{ref}} \right) dB$$
 [2.1]

where P_{ref} is equal to 20 µPa or the equivalent in other units, the threshold of hearing.

SPL needs to be measured or predicted in the frequency range from 50 to 10,000 Hz; where noise lower than 50 Hz is usually not generated by aircraft and imposes little or none annoyance on people relative to other frequencies (See Figure 2.1), and noise higher than 1,000 Hz dissipates very quickly in the atmosphere and is thus dismissed as having any weight. This frequency range constitutes for 8 octave bands and by FAA definition dividing each band into 3 provides for a satisfactory resolution, resulting in a

total of 24 1/3 octave bands. The measurements should preferably be made at the midrange frequency of each 1/3 octave band.

2.2 Calculation Procedures

There are four major steps to deriving the final value of the *EPNL*. Note that this is a spatial function as well as a function of time and that these procedures need to be repeated for every point of interest.

2.2.1 Annoyance Correction

The 24 1/3 octave bands of sound pressure levels are converted to perceived noisiness by means of a noy table (See Appendix A). This is to account for the annoyance response of the human hearing system as a function of frequency. The following graph shows this response. This graph is for a noy value of 1, with respect to the 1,000 Hz midrange frequency. What this graph shows is the relative annoyance to humans over the frequency spectrum from 50 to 10,000 Hz. For instance, a sound

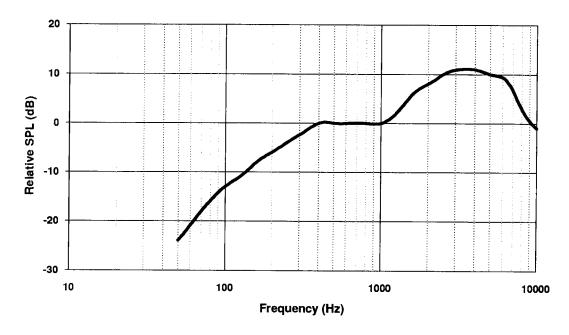


Figure 2.1 A graph of relative annoyance with respect to noise frequency.

pressure of 29 dB at 4,000 Hz has the same adverse effect or annoyance on a person as a sound pressure of 53 dB at 100 Hz.

The noy values for each band are then combined and converted to instantaneous perceived noise levels, PNL(k). The following formula is used to combine the noy values:

$$N(k) = n(k) + 0.15 \left[\sum_{i=1}^{24} n(i,k) \right] - n(k)$$

$$= 0.85 n(k) + 0.15 \sum_{i=1}^{24} n(i,k)$$
[2.2]

where n(k) is the largest of the 24 values of n(i,k), N(k) is the total perceived noisiness, i indicating the frequency band and k the time instant; whereas convention dictates that k indicates frequency. To calculate the perceived noise level or PNL(k), the following formula is used:

$$PNL(k) = 40.0 + \frac{10}{\log 2} \log N(k)$$
 [2.3]

This formula is an approximation to the SPL-Noy relation at the 1,000 Hz frequency (See Appendix A). That is: First transform the sound pressure levels in each frequency band into Noys, combine them and then transform that value back into an equivalent sound pressure level at the 1,000 Hz frequency, and designate as the perceived noise level.

2.2.2 Tone Correction

A tone correction factor, C(k) is calculated for each spectrum to account for the subjective response to the presence of spectral irregularities. This is a rather lengthy procedure consisting of ten steps.

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Step 1. Starting with the annoyance corrected sound pressure level in the 80 Hz 1/3 octave band (band number 3), calculate the changes in sound pressure level (or slopes) in the remainder of the 1/3 octave bands as follows:

Step 2. Encircle the value of the slope, s(i,k), where the absolute value of the change in slope is greater than five; that is, where

$$\left|\Delta s(i,k)\right| = \left|s(i,k) - s(i-1,k)\right| \rangle 5$$
 [2.5]

Step 3.

- If the encircled value of the slope s(i,k) is positive and algebraically greater than the slope s(i-1,k) encircle SPL(i,k).
- ii) If the encircled value of the slope s(i,k) is zero or negative and the slopes(i-1,k) is positive, encircle SPL(i-1,k).
- iii) For all other cases, no sound pressure level value is to be encircled.
 - Step 4. Compute new adjusted sound pressure levels SPL'(i,k) as follows:
- For non-encircled sound pressure levels, let the new sound pressure levels equal the original sound pressure levels, SPL'(i,k) = SPL(i,k).
- ii) For encircled sound pressure levels in bands 1 to 23 inclusive, let the new sound pressure level equal the arithmetic average of the preceding and following sound pressure levels:

$$SPL'(i,k) = \frac{1}{2}[SPL(i-1,k) + SPL(i+1,k)]$$
 [2.6]

iii) If the sound pressure level in the highest frequency band (i=24) is encircled, set the new sound pressure level in that band equal to:

$$SPL'(24,k) = SPL(23,k) + s(23,k)$$
 [2.7]

Step 5. Recompute new slopes s'(i,k), including one for an imaginary 25-th band, as follows:

$$s'(3,k) = s'(4,k)$$

 $s'(4,k) = SPL'(4,k) - SPL'(3,k)$
:
 $s'(i,k) = SPL'(i,k) - SPL'(i-1,k)$
:
:
 $s'(24,k) = SPL'(24,k) - SPL'(23,k)$
 $s'(25,k) = s'(24,k)$ [2.8]

Step 6. For i from 3 to 23 compute the arithmetic average of the three adjacent slopes as follows:

$$\bar{s}(i,k) = \frac{1}{3} [s'(i,k) + s'(i+1,k) + s'(i+2,k)]$$
 [2.9]

Step 7. Compute final 1/3 octave band sound pressure levels, SPL''(i,k), by beginning with band number 3 and proceeding to band number 24 as follows:

Step 8. Calculate the differences, F(i,k) between the original sound pressure level and the final background sound pressure level as follows:

$$F(i,k) = SPL(i,k) - SPL''(i,k)$$
 [2.11]

and note only values equal to or greater than one and a half.

Step 9. For each of the relevant 1/3 octave bands (3 to 24), determine tone correction factors from the sound pressure level differences F(i,k) and Table 2.1.

Step 10. Designate the largest of the tone correction factors, determined in Step 9, as C(k). This tone correction factor is added to the perceived noise level to obtain tone corrected perceived noise level, PNLT(k), using this simple formula:

$$PNLT(k) = PNL(k) + C(k)$$
 [2.12]

The maximum value of the tone corrected perceived noise level, PNLTM, is then determined from the time history; $PNLTM = PNLT(k)|_{MAX}$

2.2.3 Duration Correction

A duration correction factor, D, is computed by integration under the curve of tone corrected perceived noise level versus time, according to the following formula:

$$D = 10 \log \left[\left(\frac{1}{T} \right) \int_{t(1)}^{t(2)} \operatorname{antilog} \frac{PNLT}{10} \right] dt - PNLTM$$
 [2.13]

where T is a normalizing time constant, PNLTM is the maximum value of PNLT, and t(1) and t(2) are the limits of the significant noise time history.

Since *PNLT* is calculated from measured or predicted values of *SPL*, there will be in general no obvious equation for PNLT as a function of time. Consequently, the equation shall be rewritten with a summation sign instead of an integral sign as follows:

Frequency $f(Hz)$	Level Difference F (dB)	Tone Correction C (dB)
	$1.5 \le F < 3$	F/3-0.5
$50 \le f \le 500$	$3 \le F < 20$	<i>F</i> /6
	20 ≤ <i>F</i>	31/3
	1.5 ≤ <i>F</i> < 3	2 F/3 -1
$500 \le f \le 5,000$	$3 \le F < 20$	F/3
	$20 \le F$	62/3
	1.5 ≤ <i>F</i> < 3	F/3-0.5
$5,000 \le f \le 10,000$	$3 \le F < 20$	F/6
	$20 \le F$	31/3

Table 2.1 Tone Correction Factor

$$D = 10 \log \left[\left(\frac{1}{T} \right) \sum_{k=0}^{d/\Delta t} \Delta t \cdot \operatorname{antilog} \frac{PNLT(k)}{10} \right] - PNLTM$$
 [2.14]

This equation can be further simplified by taking the Δt and T outside the summation, resulting in:

$$D = 10 \log \left[\sum_{k=0}^{d/\Delta t} \operatorname{antilog} \frac{PNLT(k)}{10} \right] - PNLTM + 10 \log \left(\frac{\Delta t}{T} \right)$$
 [2.15]

where Δt is the length of the equal increments of time for which PNLT(k) is calculated and d is the time interval to the nearest second during which PNLT(k) is within a specified value, h, of PNLTM. This is depicted in Figure 2.2.

To obtain a satisfactory time history of the perceived noise level, Δt should be set to 500 ms, or a shorter approved limit. Other constants are

$$T = 10 \text{ s}$$

 $h = 10 \text{ dB}$

Using these values the equation for D becomes:

$$D = 10 \log \left[\sum_{k=0}^{2d} \operatorname{antilog} \frac{PNLT(k)}{10} \right] - PNLTM - 13$$
 [2.16]

2.2.4 Final Objective

The effective perceived noise level, *EPNL*, is determined by the algebraic sum of the maximum tone corrected perceived noise level and the duration correction factor:

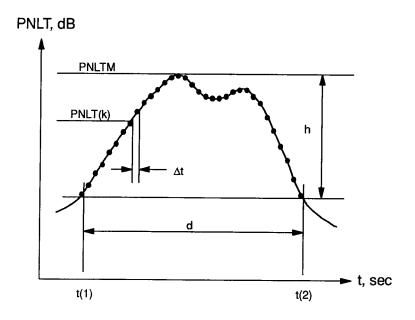


Figure 2.2. The noise time history and the related terms.

$$EPNL = PNLTM + D ag{2.17}$$

and represents the total subjective effects of an airplane flyover.

3. FAR NOISE CERTIFICATION REQUIREMENTS

Noise limits for supersonic transport category airplanes are specified in section §36.301 of FAR part 36 [Ref. 2]. The specifications are short, the noise levels should be "...reduced to the lowest levels that are economically reasonable, technologically practicable, and appropriate..."[Ref. 2]. However, FAA has stated that future HSCT aircraft will be required to adhere to Stage 3 limits, which are currently required of all new subsonic airplanes.

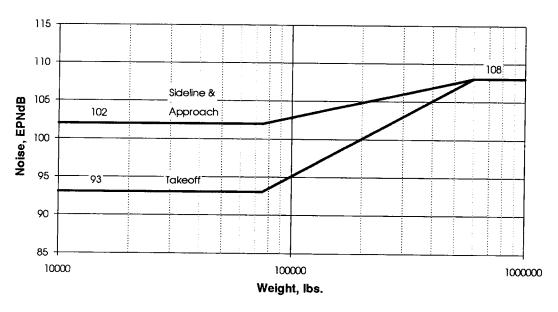


Figure 3.1 Stage 2 certification requirements.

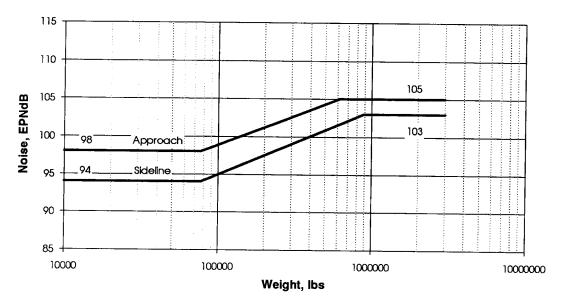


Figure 3.2 Stage 3 certification requirements for sideline and approach.

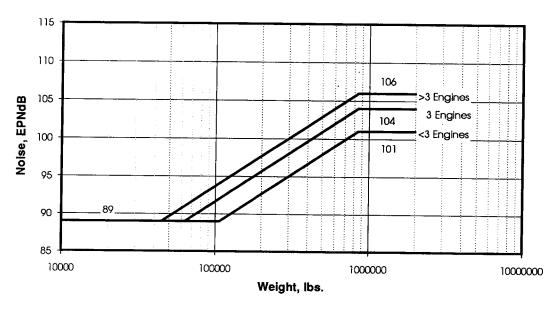


Figure 3.3 Stage 3 certification requirements for takeoff.

Figure 3.4 shows the reference points for approach, takeoff and sideline measurements. These points are the same for all stages. The points are located as follows:

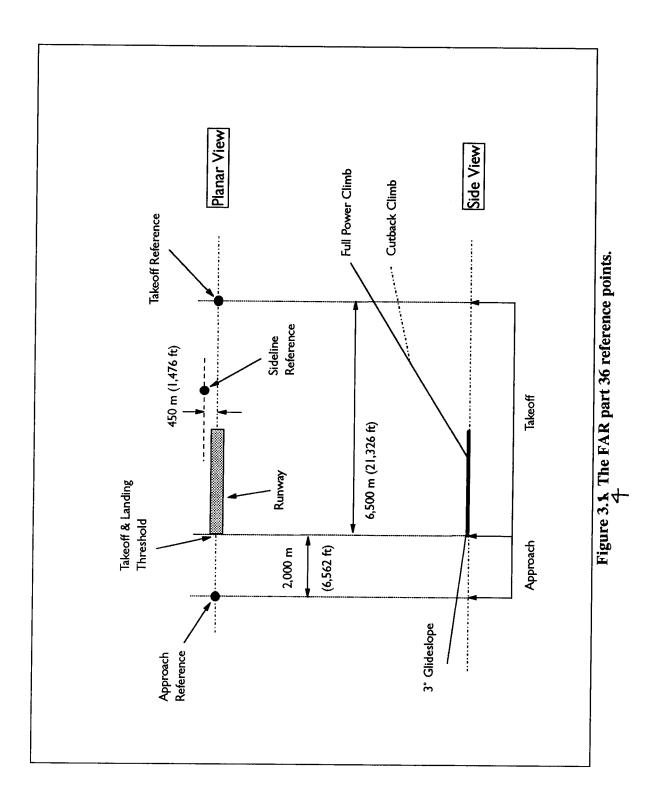
- For takeoff, at a point 6,500 meters from the start of the takeoff roll on the extended centerline of the runway.
- For approach, at a point 2,000 meters from the threshold on the extended centerline of the runway
- For sideline, at the point on a line parallel to and 450 meters from the extended centerline of the runway, where the noise level after liftoff is greatest.

3.2 Stages

Figures 3.1 through 3.3 show stage 2 and stage 3 requirements versus weight, for takeoff, sideline and approach. See Appendix B for more specifics. Stage 1, the original limits, are not included since they are no longer applicable. These requirements are not all that strict though, since tradeoffs are allowed. The noise level limits at one or two of the measuring points may be exceeded, if:

- The sum of the exceedances is not greater than 3 EPNdB,
- No exceedance is greater than 2 EPNdB, and
- The exceedances are completely offset by reductions at the other points.

FAA is also considering lessening the restrictions on supersonic transports in terms of takeoff requirements [Ref. 4], which are more restrictive than FAR pt. 25 for the purpose of noise certification. Noise reduction through advanced takeoff procedures has been studied [Ref. 3], and shown to be somewhat effective. This study used the ANOPP noise prediction program along with the FLOPS flight optimization system.



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- 2. DOT/FAA Noise Standards: Aircraft Type and Airworthiness Certification, FAR Part 36, January 1, 1990.
- 3. **E.D. Olson**, Advanced Takeoff Procedures for High-Speed Civil Transport Community Noise Reduction, <u>SAE Technical Paper Series</u>, No. 921939, 1992.
- 4. Civil Supersonic Aircraft Noise Type Certification Standards and Operation Rules; Notice of Proposed Rulemaking. Federal Register Vol. 55, No. 104, Wednesday, May 30, 1990. [J1 A2]

APPENDIX A

This appendix contains a table of the annoyance values for a range of frequency and sound pressure level.

Frequency, Hz

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SPL, dB

Frequency, Hz

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9300	2.63	2 84	1 8	300	346	374	200	26.0	9 9	8	2 2	47.0	10.0	5	44	9	7.39	7.92	8.49	9.06	9.74	10.4	11.2	12	12.8	13.8	14.7	15.8	16.9	18.1	19.4	20.8	22.3	23.9	25.6	27.4	29.4	31.5	33.7	36.1	38.7	415	
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4000	9.00	323	346	377	3.97	4 26	3	8 8	20,4	100	000	2	1 6	200	3 2	7	8.49	89	9.74	10.4	11.2	12	12.8	13.8	14.7	15.8	16.9	18.1	19.4	20.8	22.3	23.9	25.6	27.4	29.4	31.5	33.7	36.1	38.7	41.5	44.4	47.6	
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2500	2.81	302	3.23	3.46	3.71	3.97	4.26	45	8	200	3.64	0 0	3	5 6	9 6	3	7.92	9.49	80.6	9.74	10.4	11.2	12	12.8	13.8	14.7	15.8	16.9	18.1	19.4	20.8	22.3	23.9	25.6	27.4	29.4	31.5	33.7	36.1	38.7	41.5	44.4	ļ
2000	2.45	2.63	2.81	3.02	3.23	3.46	3.71	3.97	4.26	3 4	8 8	201	2 64	5 6	0.0		6.9	3	7.92	8.49	9.09	9.74	10.4	11.2	12	12.8	13.8	14.7	15.8	16.9	18.1	19.4	20.8	22.3	23.9	25.6	27.4	29.4	31.5	33.7	36.1	38.7	
1600	2.14	2.29	2.45	2.63	2.81	3.02	3.23	3.46	371	6	4 26	4 56	3 8	200	1 2	0 0	50.0	4	6.9	7.38	7.92	8.49	60.6	9.74	10.4	11.2	12	12.8	13.8	14.7	15.8	16.9	18.1	19.4	20.8	22.3	23.9	25.6	27.4	29.4	31.5	33.7	1
1250	1.62	1.74	1.87	2	2.14	2.3	2.46	264	2.83	8	375	348	2 2		1 00	27.	8.3	4.82	5.28	2.66	90.9	6.5	96.9	7.46	8	8.57	9.19	9.85	10.6	1.3	12.1	5	13.9	14.9	16	17.1	16.4	19.7	21.1	22.6	24.3	56	
1000	1,41	1.52	1.62	1.74	1.87	2	2.14	2.3	2.46	254	8	303	3.25	3.48	2 2	3	4 6	8	4.59	4.82	5.28	5.66	90.9	6.5	96.9	7.46	80	8.57	9.19	9.82	10.6	1:3	12.1	5	13.9	14.9	16	17.1	18.4	19.7	21.1	22.6	;
800	1.41	1.52	1.62	1.74	1.87	2	2.14	2.3	2.46	264	283	388	3.25	3.48	2 2	;	* 6	4.63	4.59	4.92	5.28	5.66	90.9	6.5	96.9	7.46	8	8.57	9.19	9.82	10.6	E. E.	12.1	<u>5</u>	13.9	14.9	16	17.1	18.4	19.7	21.1	22.6	
630	1.41	1.52	1.62	1.74	1.87	2	2.14	2.3	2.46	264	2.83	3.03	3.25	348	9 73		1 00	27	63.4	4.92	5.28	5.66	90.9	6.5	96.9	7.46	æ	8.57	9.19	9.82	10.6	1.3	12.1	5	13.9	14.9	16	17.1	18.4	19.7	21.1	22.6	7
200	1.41	1.52	1.62	1.74	1.87	2	2.14	2.3	2.46	264	2.83	303	3.25	3.48	3 73	+	1 0	2 2	86.4	78.4	5.28	5.66	90.9	6.5	96.9	7.46	8	8.57	9.19	9.82	10.6	E.	12.1	3	13.9	14.9	9	17.1	18.4	19.7	21.1	22.6	
400	1.41	1.52	1.62	1.74	1.87	2	2.14	2.3	2.46	2.64	2.83	3.03	3.25	3.48	3.73	T	1 02 4		8 5	4.82	5.28	2.66	90.9	6.5	98.9	7.46	8	8.57	9.19	9.82	10.6	E.	12.1	9	13.9	4,9	9	17.1	18.4	19.7	21.1	22.6	0 70
315	1.24	1.33	1.42	1.53	1.64	1.76	1.89	2.03	2.17	2.33	2.5	2.69	2.88	3.1	3.32	2.57	883	1	7	4	4.73	2.08	5.45	5.85	6.27	6.73	7.23	7.75	8.32	8.93	9.59	10.3	=	÷	12.7	13.6	14.6	15.7	16.9	18.1	19.4	20.8	7 00
250	1.08	1.16	1.25	1.34	1.45	1.56	1.68	1.8	<u>ş</u> .	5.09	2.25	2.42	2.61	2.81	3.03	3.26	354	1	0 9	3 1	4.38	4.71	2.07	5.46	5.88	6.33	6.81	7.33	6:	3.5	9.15	9.82	10.6	11:3	12.1	13	13.9	14.9	9	17.1	18.4	19.7	2.
200	0.91	F	1.08	1.17	1.26	1.36	1.47	1.58	1.71	1.85	2	2.15	2.33	2.51	2.71	2 89	3.16	1	4 6	8 8	86.5	6,4	4.62	5.01	5.41	5.84	6.31	6.81	98.7	3, 1	8.57	9.19	9.85	10.6	E.	12.1	<u></u>	13.9	14.9	16	12.1	18.4	407
160	0.74	0.82	6:0	-	1.08	1.17	1.26	1.36	1.47	1.58	1.71	1.85	7	2.15	2.33	254	2.71	8	2 4	5 6	£ 6	3.69	3.99	4.3	4.64	5.01	5.41	2.84	6.31	50 5	8 2	3, 1	6.5	6 6	9.85	10.6	1.3	12.1	5	13.9	14.9	16	
125	0.55	0.61	0.67	0.74	0.82	6.0	F	1.08	1.18	1.28	1.35	1.5	1.63	1.77	1.92	2.08	2.26	3,45	2 2	3 2	80.7	3.12	3.39	3.68	98.	.33	69.4	30 5	76.5	S C	200	3 5	8 8	87.0	8	9.76	10.6	7.3	12.1	2	13.9	14.9	9
ŝ	0.42	0.46	0.52	0.58	0.65	0.72	0.8	6.0	=	1.09	1.15	1.29	1.4	1.53	1.66	181	1.97	3 15	2 6		7.3	7	3.01	3.28	3.57	3.98	2 5	9 3	5.07	0.4°	\$ 5	9 5	2 5	8 8	33.	8.07	985	10.6	1.3	12.1	5 5	13.8	4.9
88	0.26	0.3	98	0.38	0.43	0.49	0.55	0.62	0.7	0.79	0.89	-	1.09	1.18	1.29	-	1.53	4 66	1 2		2	61.2	2.34	4.5	2.77	10.01	200	20.0	8 5	4.63	2 2	5 2	2	g (9.40	89.	99.7	8.33	9.07	9.87	2		12.7
63	0.12	0.14	0.16	0.19	0.22	0.26	0.3	0.34	0.39	0.45	0.51	0.59	0.67	0.77	0.87	=	Ξ	12	. 6		2	اِه	2 5	1.92	2.11	7.32	66,3	27.7	9.07	10.0	, ,	3 4	9 0	1.09)S' (3	6.48		7.81	8.57	9.41	10.3	69
90	+	+		1	5.	0.12	0.4	0.17	0.21	0.25	0.3	0.34	0.39	0.45	0.51	0.59	0.67	0.77	0.87	7	-		77	3	24.	8 8	20.5	20.0	2,45	2 72	2/6	5 6	3.32	600	0 5	6.49	96	5.48	90.9	9',	141	8.19	60.6
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 $\mathrm{Sb}\Gamma$ ' $\mathrm{q}\mathrm{B}$

Frequency, Hz

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2000	44.4	47.6	2	547	58.6	2 69	27.2	7.70	7	7,7	82.7	98.6	<u>9</u> 0	102	109	117	125	±	4	<u>7</u>	165	12	188	g	217	233	249	267	286	307	329	352	377	404	433	464	497	533	571	611	655	702	752
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1000	26	27.9	29.9	33	34.3	36.8	2 00	40.00	1 2	200	6.0	25	55.7	29.7	8	9.89	73.5	78.8	84.4	90.5	97	₽	Ξ	119	128	137	147	158	169	181	호	208 208	523	33	556	274	294	315	338	362	388	416	446
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315	77	25.8	27.7	29.7	31.8	34.2	36.7	39.4	42.2	45.0	70 7	ç ç	7 :	20.7	29.7	정	9.89	73.5	78.8	4.4	90.5	97	\$	Ξ	119	128	137	147	28	8	<u>.</u>	<u>a</u>	8	223	738	556	274	28	315	338	362	88	416
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200	21.1	22.6	24.3	26	27.9	59.9	32	34.3	36.8	30.4	42.2	4 5	200	0.0	76	20 2	28.7	8	9.89	73.5	78.8	4.4	90.5	26	ş	Ξ	119	128	137	47	2 3	2 3	<u> </u>	3 3	802	223	330	556	274	8	382	88	362
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125	17.1	18.4	19.7	21.1	22.6	24.3	56	27.9	29.9	8	34.3	98	9 8	4.00	7 2 2	50.0	0.0	25	55.7	29.7	क्र	9.89	73.5	78.8	4.4	90.5	6	호	=	61.	2 5	2 ;	4 2	2 3	2 3	<u>.</u>	<u>ş</u>	8	223	539	256	2/4	\$ 2
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SPL, dB

Frequency, Hz

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008		478	512	549	588	93	929	724	9//	833	ģ	, H	3 3	1024	1098	1176	1261	1351	1448	1552	1664	1783	191	2048
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400		478	512	548	588	089	929	724	776	832	88	955	į	42	<u>8</u>	1176	1261	1351	1448	1552	1664	1783	191	2048
315		446	478	512	549	288	089	929	724	776	832	68	į	cc S	1024	1098	1176	1261	1351	1448	1552	1664	1783	1911
250		416	446	478	512	549	288	930	929	724	776	833	Š	8	955	1024	1098	1176	1261	1351	1448	1552	1664	1783
200		88	416	446	478	512	549	288	630	929	724	776	66	3 3	£	955	1024	1098	1176	1261	1351	1448	1552	1664
160	1000	33	362	388	416	446	478	512	549	288	930	929	107	1, 2,	*	832	891	955	1024	1098	1176	1261	1351	1448
125	177	25	88	362	388	416	446	478	512	549	288	အ	A7A		<u>*</u>	9//	832	891	955	1024	1098	1176	1261	1351
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63	200	3	3	556	274	285	315	338	362	388	416	446	478	1	210	8	288	630	929	724	776	832	891	955
20	auc	3 8	227	88	528	274	294	315	88	362	388	416	446	12,	9	212	22	88	930	929	724	776	832	891
السب	120	9 9	3	131	132	<u> </u>	<u>\$</u>	135	136	137	138	139	140	3	: :	142	5	1 4	145	146	147	148	149	150

SPL, dB

APPENDIX B

This appendix derives and explains how to obtain the FAR 36 noise limits for aircraft which weights are in the nonlinear range of the FAR 36 specifications. Figures 3.1 through 3.3 are presented using a logarithmic scale, which makes it hard to read off values. Let's first look at how FAA presents the requirements. As an example, the stage 2 requirements for takeoff reads as follows:

"...108 EPNdB for maximum weights of 600,000 pounds or more, reduced by 5 EPNdB per halving of the 600,000 pounds maximum weight down to 93 EPNdB for maximum weights of 75,000 pounds and less."

[Reference 2]

Essentially, what is given is a starting and ending point. In between is a non-linear region defined by a single constant: reduction in noise per halving of the weight. This is in effect a half-life constant, T, as for radioactive materials. Recall that the instantenaeous rate of decay of a radioactive substance is proportianal to the amount present, or

$$\frac{dy}{dt} = rt ag{B.1}$$

where y is the mass, t is time and r is a neagtive constant. The general soution is,

$$y = Cb^{rt}$$
 [B.2]

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where C and b are positive constants. By introducing the half-life of a substance, T, this equation becomes,

$$y = C\left(\frac{1}{2}\right)^{t/T}$$
 [B.3]

which for aircraft noise translates into,

$$W = C\left(\frac{1}{2}\right)^{-\left(\frac{EPNL}{T}\right)} = C(2)^{\frac{EPNL}{T}}$$
 [B.4]

where W is the weight of the aircraft and EPNL is the noise certification level required for that aircraft. Note that the negative sign on the exponent comes from the fact that EPNL is a positive constant representing a reduction in noise levels.

Since the noise level is what we want, equation B.4 can be rewritten as,

$$EPNL = \frac{ln\left(\frac{W}{C}\right) \cdot T}{ln 2}$$
 [B.4]

where the constants required in the equation can be looked up in the following table, along with the region that the equation is valid in, represented by W_o and $W\infty$, as well as the relevant noise level at that weight.

Table B.1 Constants for Noise Requirement Calculations

			C	T	Wo	W∞
	Appr. &	Sidel.	$3.33*10^{-11}$	2	75,000@102	600,000@10
Stage 2	Takeoff		$1.89*10^{-01}$	5	75,000@93	600,000@10
		>3 Eng.		4	44,673@89	850,000@10
Stage 3	Takeoff	=3 Eng.	$1.27*10^{-02}$	4	63,177@89	850,000@10
		<3 Eng.	$2.13*20^{-02}$	4	106,250@89	850,000@10
	Sideline		$6.82*10^{-07}$	2.56	77,200@94	882,000@10
	Approach		$1.68*10^{-08}$	2.33	77,200@98	617,300@10